ANSYS Mechanical 2022R2 新功能介绍

新科益系统与咨询 (上海) 有限公司



Intuitively designed, integrated and dependable solver technology

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B

Engineering Data

lodal

Geometry

Model

Setup

Solution

Results



New intuitively designed, and customizable toolbar for Add-ons

- ✓ Access multipurpose workflows quickly and efficiently
- ✓ On-Off button for load/unload
- Training content under Ansys Help and Ansys ALH

Robust workflow improvements of MSUP Harmonic and MSUP Transient

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Harmonic Response

Engineering Data

Geometry

Model

Setup

Solution

Results

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- Reduce overall disk space requirement by 4X
- Solver speed performance improvement
- ✓ Data mapping time from structural to acoustics harmonic dropped by 50%

Increased exposure of Coupled Field analysis

- Enables easier workflows for sensor design, MEMs devices and actuators
- ✓ ability to model piezoelectric and acoustic degrees of freedom
- Includes a library of commonly used piezoelectric materials used in sensor design



Table of Contents

- <u>Unified Contact Detection option</u>
- <u>Accurate wear modeling and</u> <u>Fretting Fatigue life prediction</u>
- Toolbar for Add-ons
- <u>MSUP Harmonic and MSUP</u> <u>Transient workflow</u>
- Exposure of Hybrid Parallel
- <u>Coupled Field analysis exposure</u>
- <u>Hyper viscoelastic materials in</u> <u>Linear Perturbation</u>

- Support for AMD GPUs
- Default number of cores
- **SMART enhancements**
- <u>Composite post-processing</u> <u>functionalities</u>
- Material Support for the LS-DYNA Solver
- <u>Element embedding workflow</u>
- <u>Co-Simulation of Aqwa, Rigid Dynamics</u> and AeroDyn
- Structural Optimization



New Feature:

A new unified contact detection option

Value Provided:

Improves solution robustness for non-smooth contact applications



A Unified Contact Detection Method

- A new unified contact detection option (KEYOPT(4)=5) is exposed in 2022R2 (available in Mechanical) which combines three individual contact detection methods together: Gauss point (KEYOPT(4)=0), nodal point (KEYOPT(4)=2), surface projection (KEYOPT(4)=3).
- It adds more contact constraint points at contact interface and results in much less penetration and less mesh sensitivity.
- It robustly solves non-smooth contact problems with different contact scenarios which were generally solved by explicit solvers in the past.







New Feature:

Accurate wear modeling and Fretting Fatigue life prediction

Value Provided:

Automatically scales the wear process so that a few simulation cycles can represent several hundred real cycles, thus enabling accurate prediction of fretting fatigue life



Automatic Wear Scaling-Fretting Fatigue

- Wear & Fretting fatigue are slow processes- simulation for the whole process is prohibitively expensive
- New feature to automatically scale the wear to simulate large number of cycles
- Accurate Wear modeling + nCode=Accurate fretting life prediction

Applied		With wear			Without wear			
sliding amplitude (mm)	Physical cycles	Calculated life (repeats)	Calculated life (cycles)	Experimental life (cycles)	Physical cycles	Calculated life (repeats)	Calculated life (cycles)	
0.184	1,385	30.08	41,661	43,983	6	2,398	14,388	
0.147	1,485	27.22	10,422	37,717	6	2,278	13,688	
0.073	1,888	20.71	39,100	28,800	6	2,972	17,832	

Life prediction with new autoscaling + nCode match experiments –both qualitatively (wear increases fretting life as see in experiments) and quantitatively





Automatic scaling enables-6 simulation cycles = 1385 real cycles



New Feature:

New Toolbar for Ansys Mechanical Add-ons

Value Provided:

- Access multipurpose workflows quickly and efficiently
- On-Off button for load/unload
- Training content under Ansys Help and Ansys ALH



Ansys Mechanical Add-ons : Value Messaging

Owned, Developed and Managed by Ansys

Part of installation package of Ansys Mechanical Product/Application

Aligned with targeted Industry or application requirements



Ansys Mechanical Add-ons : 2022 R2 – Enhancements

Easy access & Intuitive design

Available under Ansys Mechanical Toolbar

Not loaded by default

On-Off button for load/unload

Sufficient training content under Ansys Help and Ansys ALH

Usage/Popularity can be tracked under APIP Program

License controlled



Ansys Mechanical Add-ons : 2022 R2 – 14 Different Add-ons

Fatigue	DesignLife Fatique	NVH Toolkit NVH	Forced Response Turbomachinerr	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic ! oads	LPBF DED Sintering Distorti Process Process Process Compensa Additive Manufacturing	on ation Rigid Dynamics
NVH	DesignLife Fatigue	NVH Toolkit NVH						on ation Variable Motion Load Load Transfer Rigid Dynamics
Turbomachinery	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distorti Process Process Compensi Additive Manufacturing	on ation Variable Motion Load Load Transfer Rigid Dynamics
Explicit	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distorti Process Process Compense Additive Manufacturing	on ation Variable Motion Load Load Transfer Rigid Dynamics
Mechanical Toolkit	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distorti Process Process Compensi Additive Manufacturing	on ation Variable Motion Load Load Transfer Rigid Dynamics
Hydrodynamic Loads	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distorti Process Process Compens Additive Manufacturing	ion ation Variable Motion Load Load Transfer Rigid Dynamics
Additive Manufacturing	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distortion Process Process Process Compensation	n ion n Kigid Dynamics
Rigid Dynamics	DesignLife Fatigue	NVH Toolkit NVH	Forced Response Turbomachinery	Drop Test Explicit	Bolt Statistics on Tools Structures Mechanical Toolkit	Hydrodynamic Offshore Pressure Hydrodynamic Loads	LPBF DED Sintering Distorti Process Process Compense Additive Manufacturing	on Variable Motion Load Load Transfer Rigid Dynamics



Ansys Mechanical Add-ons : 2022 R2 – Design





NVH Toolkit Add-on



NVH Toolkit Add-on: MAC UI enhancements

- Addition of the 3D MAC Table.
- Added support for rst-rst file comparison without model rotation. This option is enabled to identify
 modal changes in computational models if design changes are introduced (e.g. change in the
 materials, mesh, contact hypotheses...), but not changes in terms of global positioning. The rest of the
 MAC Calculator functionalities continue to be applicable.
- Mode Preview implemented in the Frequency Worksheet to be able to animate modes before solving the MAC Calculator.



-	File 1 Options						
	File Type	rst					
	File	D:\simple\simple_files\dp0\SYS\MECH\file.rst					
-	File 2 Options						
	File Type	rst					
	File	D:\simple\simple_files\dp0\SYS\MECH\MA					
3	MAC Calculation Options						
Ì	Node Matching Absolute Tolera	0.01 m					
	Nearest Node match	Yes					
	Restrict to Nodal Named Selection	No					
	Degrees of Freedom	All Structural DOFs					
3	Mode Pairing Options						
	Pair Modes	No					
	MAC Limit	0.9					
	Frequency Tolerances	Program Controlled					
]	Optimization Result						
	Alpha (Frequency term)	1					
	Objective Function (f)	0.0000E+00					





NVH Toolkit Add-on: COMAC Calculation

• Coordinate Modal Assurance Criterion (COMAC) has been added as an additional output of MAC. COMAC is computed for each pair of matched Degrees of Freedom

Nodes (File 1)	Nodes (File 2)	COMAC (UX)	COMAC (UY)	COMAC (UZ)	
169	6	0.0828	0.5126	0.0194	
173	7	0.0438	0.5114	0.0117	
180	3	0.0461	0.5114	0.0124	
184	2	0.0871	0.5126	0.0232	
188	1	0.0000E+00	0.0000E+00	0.0000E+00	
194	5	0.0000E+00	0.0000E+00	0.0000E+00	
195	8	0.0733	0.2718	7.9320E-03	
201	4	0.0686	0.2718	7.8081E-03	

- As MAC, COMAC is bounded between 0 and 1.
- With COMAC, users can identify the potential sources of low MAC (associated with DOF pairs with low COMAC) or the reasons for a high MAC (DOF pairs with high COMAC).



NVH Toolkit Add-on: FRF Calculator

• The Frequency Response Function (FRF) Calculator is a brand-new post object that allows users to compute the relationship between a generalized input force and a generalized output displacement as a function of frequency:





NVH Toolkit Add-on: FRF Calculator, connection with UNV data

- UNV Data has been incorporated into the FRF Calculator to be able to analyze experimental FRFs and compare them with computational FRFs.
- Fully defined model orientation is available (by Coordinate System, Rigid Body Transformation or 3 Node Alignment).

FRF Definiton						0	000	0.050	0.10	1(m)	X
Frequency Minimum	0 Hz										
Frequency Maximum	2000 Hz	1					0.025		0.075		
Frequency Interval	1 Hz	provide a second									the second s
Nodes Definition	Manual	FRF Works	sheet								▼ # □
UNV Data		al			-		-				
include UNV Data	Yes				422 4		HH I				
ile Type	unv	Lise in FR	E Damoed Fred (Hz)	Dampine A	Show	Output Node	Dutrut DOF	Innut Node	Innut DOF	Name	
Inits	Dataset 164 (UNV File)		627.40	2.00		216269	IN	222491	F7	(M2162691/67212	
ile	D:\\\FR		639.19	2.00		272047	117	272099	FY	1/7/272047//EV/2	
Orient By	Rigid Body Transformation		869.49	200		69012	LIZ.	272099	FY	LIZI690121/FY127	
Translation	0.097 m		876.18	2.00		461402	LIY.	272099	FY	LIV14614020FY12	
Translation	0.179 m		1589.17	2.00	-	101102		212000		o reconcept recent	
Z Translation	-0.001 m		1589.43	2.00							
Rotation X Axis	0		1664.91	2.00							
Rotation Y Axis	0		1684.89	2.00							
Rotation Z Axis	1		1685.69	2.00							
Rotation Angle	90 *		1827.21	2.00							
	18103		1959.54	2.00							
			2067.91	2.00	_				44		
			2068.52	2.00	Chan	Outrust Made	Outrue DOE	Jamest Manda	Inna DOF	Mama	
			2119.72	2.00	STROW	E 40	OupurDor	Input Node	Input Dor	I TRE 40 ID ARCON	^^
			2168.96	2.00		543	02	560	FT	U2(549/FT(560)	
			2190.78	2.00	H	54/	02	560	PT .	U2(547)/FT(560)	
			2194.81	2.00	H	546	02	000	FT N	U2(548)/F1(560)	
			2194.82	2.00	H	545	02	060	FT	U2(545)/F1(560)	
			2279.58	2.00	H	10	UX UX	560	FT	03(10)/FT(560)	
			2280.02	2.00	H	61	UY	560	FY	01(61)/11(500)	
			2282.43	2.00		71	UT	560	FY	01(/1/+1(560)	
			2335.70	2.00		543	02	560	FY	02(543/17(560)	
			2384.49	2.00	H	544	02	560	FY	UZ[544//FY[560]	
			2408.93	2.00	H	546	02	560	FY	02(546)/17(560)	
		~	2102.01	```````` ```	H	550	02	560	FT	02(550)/FT(560)	



NVH Toolkit Add-on: FRAC Calculator

- A Frequency Response Assurance Criterion (FRAC) Calculator has been incorporated into the FRF Calculator as an additional tool to measure correlation between FRFs.
- FRAC can be computed between two computational FRFs or between a computational and a experimental FRF.
- An automatic algorithm that matches node pairs between the experimental and computational models has been implemented, and users are be able to quickly compare the same FRFs in both models.



1#1

Show	FRF1	FRF2	Name	FRAC	
~	UY[216268]/FZ[233491]	UZ(543)/FY(560) (unv)	FRAC_PAIR_1	0.0124	
~	UZ(272047)/FY(272099)	UZ(549)/FY(560) (unv)	FRAC_PAIR_2	5.3934E-03	
	UZ(69012)/FY(272099) (UZ(543)/FY(560) (unv)	FRAC_PAIR_3	0.0339	
	UY(461402)/FY(272099)	UY(21)/FY(560) (unv)	FRAC_PAIR_4	0.0569	



Forced Response Add-on



Forced Response Add-on

- The Forced Response Tool provides a comprehensive way of investigating the aeromechanics and reliability of turbomachinery blade rows.
- In conjunction with Ansys Computational Fluid Dynamics (CFD) tools, the Forced Response Tool accurately predicts the structural vibrations of industrial components. Advanced physical phenomena such as airfoil aeroelasticity, flutter, deterministic and probabilistic mistuning, as well as an array of loading and boundary conditions can be modeled and studied.
- Efficient high-fidelity modeling is achieved using a combination of core technologies such as cyclic symmetry, mode-superposition, and innovative techniques to model mistuning and aeroelasticity in a reduced space.
- This tool provides an intuitive, reliable, and efficient way to simulate turbomachinery blade rows.





DesignLife Add-on



DesignLife Add-on: Analysis from Mechanical

• Ability to insert the DesignLife analysis directly from Mechanical.





DesignLife Add-on: Harmonic Direct – Vibration Fatigue

Vibration Fatigue Load	Stress	Strain	Shell Seam Weld	Solid Seam Weld
PSD Loading, Including: - Static Offset Case - Single and Multiple Events	\checkmark	\checkmark	\checkmark	\checkmark
<u>Single Frequency Loading, Including:</u> - Static Offset Case - Single and Multiple Events	\checkmark	\checkmark	\checkmark	\checkmark
<u>Frequency Range Loading, Including:</u> - Static Offset Case - Single and Multiple Events	\checkmark	\checkmark	\checkmark	\checkmark
<u>Sine On Random Loading, Including:</u> - Static Offset Case - Single and Multiple Events	\checkmark	\checkmark	\checkmark	\checkmark



Single Frequency Vibration Load – Stress Case

• Life Result Comparison between DesignLife Add-on and Standalone:



DesignLife Add-on







DesignLife Add-on: Expose Static (Preload) Option within Time Series

- In order to define the static load case (preload)
 - Set the "Static" option to "Yes".
 - Then the input file selection will not be required.

•This sets the Static State to "Yes" or "No" within the command SetLoadCaseStaticState inside the input.dcl.

•For clarity, the name of the load will be updated to include (Static) specification.

Definition Environment Static Structural Define By Time Time 1.0 Scale Factor 1 Static No Input File Definition Absolute Path Input File View Time Series Plot View Time Series Plot No Image: Constraint of the Series Plot No	D	Details of "Time Series Load" 👻 🕈 🗖 🗙					
Environment Static Structural Define By Time Time 1.0 Scale Factor 1 Static No Input File Definition Absolute Path Input File View Time Series Plot View Time Series Plot No Imput File View Time Series Plot Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File Solution Information Details of "Time Series Load (Static)" Imput File Imput File Solution Information Definition Environment Static Structural Define By Time Time 1.0 Scale Factor Scale Factor 1 Static Yes		Definition					
Define By Time Time 1.0 Scale Factor 1 Static No Input File Definition Absolute Path Input File View Time Series Plot View Time Series Plot No Imput File View Time Series Plot View Time Series Plot No Imput File View Time Series Plot View Time Series Plot No Imput File View Time Series Plot View Time Series Load (Batic) Imput File Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Imput File View Time Series Load (Static) Im		Environment	Static Structural				
Time 1.0 Scale Factor 1 Static No Input File Definition Absolute Path Input File View Time Series Plot No		Define By	Time				
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Static No Input File Definition Absolute Path Input File View Time Series Plot View Time Series Plot No Image: Solution Group Image: Solution Information Detinition Image: Environment Static Structural Define By Time Time Time Time Time Static Yes		Scale Factor 1					
Input File Definition Absolute Path Input File View Time Series Plot View Time Series Plot No Imput File No Imput F	1	Static No					
Input File View Time Series Plot No Image: Solution Group Image: Solution Information Definition Environment Static Structural Define By Time Time 1.0 Scale Factor 1 Static Yes		Input File Definiti	on Absolute Path				
View Time Series Plot No Image: Contract of the series load (Static) Image: Contract of the series load (Static) <t< th=""><th></th><th>Input File</th><th></th></t<>		Input File					
Image: Control of the con		View Time Series F	Plot No				
Details of "Time Series Load (Static)" ✓ ↓ ↓ □ Definition		Solution Group Load Mapper Loading Event Time Series Load (Static) Materials Solution (B6) Solution Information					
□ Definition Environment Static Structural Define By Time Time 1.0 □ Scale Factor 1 Static Yes	D	etails of "Time Se	eries Load (Static)" 👻 🕂 🗆 🗙				
Environment Static Structural Define By Time Time 1.0 Scale Factor 1 Static Yes	-	Definition					
Define By Time Time 1.0 Scale Factor 1 Static Yes		Environment Static Structural					
Time 1.0 Scale Factor 1 Static Yes		Define By	Time				
Scale Factor 1 Static Yes ▼		Time	1.0				
Static Yes 🔻		Scale Factor	1				
		Static	Yes 🔻				



DesignLife Add-on: Time Series Input File (Absolute or Relative Path)

- Input File Definition:
 - Choose between "Absolute Path" or "Relative Path".
- If "Absolute Path" is used
 - The time series file will be loaded from the path selected in your local device. This means if the file is moved to a different place, the time series input file will no longer be found.
- If "Relative Path" is used
 - The local file selected will be copied to the user_files folder within the project folder. Instead of loading the file from the local path, it will load it from the user_files. So make sure, if you modify the local file, and you want your project to reflect so, to reload it again so that its updated into the user_files.

Definition	Definition						
Environment Static Structural							
Define By	Time						
Time	1.0						
Scale Factor	1						
Static	No						
Input File Definition	Absolute Path						
Input File	D:\Git_Repo\Parts\POST_Parts\ncode\NCODE_DESIGN_LIFE_APP_035\smallhistory.dat						
View Time Series Plot	No						

Static Structural
Static Structural
Time
1.0
1
No
Relative Path
\\user_files\smallhistory.dat
No



DesignLife Add-on: Post Processing Static Failure

- Life and Damage Results
- For Time Based, Stress-Life Analysis (SN), the Static Failure Damage is set to 1.234e29 in the input.dcl file consumed by nCode solver.
 - SetProperty("SNEngine_Fatigue",StaticFailureDamage,"1.234E29")
- If Static Failure is detected, Mechanical will issue the warning message 1:
 - <u>Warning message 1:</u> "Calculated alternating stress in some areas exceeded the UTS, which indicates Static Failure. For those areas, Life is set to the Static Failure Life and Damage is set to the Static Failure Damage; plotted as purple contours."



Post Processing Static Failure – Life and Damage Result

 The purple band in the legend, corresponding to the "Static Failure Life" or "Static Failure Damage" value, represents all the areas where Static Failure is reported.





DesignLife Add-on: Exposed Strain Analysis Parameters

- Geometry
 - Scoping method
 - Geometry
- Definition
 - Based on Material (automatically filled from scoped geometry)
 - Fatigue Type (Strain, read only)
 - Strength Coefficient Parametrizable
 - Strength Exponent Parametrizable
 - Ductility Coefficient Parametrizable
 - Ductility Exponent Parametrizable
 - Cyclic Strain Coefficient Parametrizable
 - Cyclic Strain Hardening Exponent Parametrizable
 - Young's Modulus
 - Poisson's Ratio
 - Tensile Ultimate Strength
 - NCode Material Type:
 - Grey Cast Iron, Nodular Cast Iron, Malleable Cast Iron, Cast Steel, Steel, Aluminum, Cast Aluminum
- Material Parameters
 - Surface Finish:
 - Polished, Ground, Machined, Poor Machined, As Rolled, AsCast.



D	etails of "Materials Assignment"	▼ ₽ □ ×				
Ξ	Geometry					
	Scoping Method	Geometry Selection				
	Geometry	1 Body				
-	Definition					
	Based on Material	Structural Steel				
	Fatigue Type	Strain				
	P Strength Coefficient	50000000 Pa				
	P Strength Exponent	-0.5				
	P Ductility Coefficient	0.675				
	P Ductility Exponent	-0.2				
	P Cyclic Strength Coefficient	100000000 Pa				
	P Cyclic Strain Hardening Exponent	0.2				
	Young's Modulus	20000000000 Pa				
	Poisson's Ratio	0.3				
	Tensile Ultimate Strength	46000000 Pa				
	nCode Material Type	Steel				
-	Material Parameters					
	Surface Finish	Polished				
-	Export Engineering Data					
	Export Engineering Data	Export				



DesignLife Add-on: Exposed Material Parameters

- Material Parameters for SN and EN
 - Surface Finish:
 - Polished, Ground, Machined, Poor Machined, As Rolled, As Cast.
 - Surface Treatment Factor
 - Can be parameterized.
 - User Surface Factor
 - Can be parameterized.

Details of "Materials Assignment" 👻 🗖 🗖 🗙					
- Geometry					
Scoping Method	Geometry Selection				
Geometry					
Definition					
Based on Material					
Fatigue Type	Strain				
Strength Coefficient					
Strength Exponent					
Ductility Coefficient					
Ductility Exponent					
Cyclic Strength Coefficien	t				
Cyclic Strain Hardening Ex					
Young's Modulus					
Poisson's Ratio					
Tensile Ultimate Strength					
nCode Material Type					
Material Parameters					
Surface Finish	Polished				
Surface Treatment Factor	1				
User Surface Factor	1				
Export Engineering Data					
Export Engineering Data	Export				



New Feature:

Weld Fatigue Workflow

Value Provided:

- Out of the box automation to handle complex weld locations
- Use expressions based on parent material thicknesses to drive different industry best practices



Seam Weld Enhancements

- Handle different complex weld locations with native automation
 - Auto switching of top & bottom faces
 - Back-to-Back welds
 - Handle multiple distinct weld locations part of one weld curve
 - Auto ignore redundant weld curves along thickness
 - Close loop welds
 - Auto retraction &/or squeeze weld &/or HAZ
 - Auto defeaturing



Туре	Continuous Seam
Source	Mesh
Modeled As	Normal and Angled
Create Using	Curves
Angled Direction	Normal
Use Worksheet	No
Curve Scoping	Geometry Selection
Weld Curve	1 Body
Definition	
Suppressed	No
Adjust Weld Height	Yes
Weld Height (Leg02) Assignment	Expression
Weld Height (Leg02) Expression	(t1+t2)/4
Creation Criteria	Width Based
Weld Width (Leg01) Assignment	Expression
Weld Width (Leg01) Expression	(t1+t2)/4
Edge Mesh Size	Default (3.0 mm)
Create HAZ Layer	Yes
HAZ Distance Assignment	Expression
HAZ Distance Expression	(t1+t2)/4
Number Of HAZ	Default (1)
HAZ Growth Rate	1.2
Generate End-Caps	Yes
Generate Named Selection	No
Intersection Tag (Beta)	
Mechanical Properties	
Material	Structural Steel
Thickness Assignment	Expression
Expression	(t1+t2)/2
Advanced	
Sharp Angle	Default (30.0°)
Connection Tolerance	Default (Program Controlled)
Smoothing	Yes
Lap Weld Angle Tolerance	5.0°

- Use expressions to drive different industry practices
 - Support for expressions
 (-,+,*,/,(,),^).
 - Available for: Weld Height, Width, HAZ Distance, Thickness





New Feature:

MSUP Harmonic and MSUP Transient workflow Performance Improvement

Value Provided:

- Reduce overall disk space requirement by 4X
- Solver speed performance improvement
- Data mapping time from structural to acoustics harmonic dropped by 50%



im Modal (A5)

Needs On Demand Expansion set to Yes in Linked Modal and no loads adding new elements.

√_{T=0} Modal (Modal) √_{T=0} Pre-Stress (None) VT=0 Modal (Modal) 🗸 🖽 Analysis Settings With MODDIR Analysis Settings Before 🗸 🖽 Analysis Settings 🖓 🔍 Pressure 🔎 Fixed Support 🔎 🔍 🖉 🔎 💭 Remote Force Solution (A6) Remote Force 🖉 📆 Solution Information 🛛 🐻 Solution (B6) ė...., 👼 Solution (C6) unition Information 🗤 🔍 Harmonic Response (B5) 🖉 📊 Solution Information 🗸 🔊 Total Deformation « Direct » 🔊 Total Deformation Analysis Settings Harmonic Response 2 (C5) 🔎 Pressure Applied By Details of "Solution (C6) - 4 □ × Details of "Solution (B6)" Remote Force Solution Solution 🖉 🐻 Solution (B6) Contraction Information Solve Process Settings Number Of Cores to Use (Beta) Solve Process Settings Number Of Cores to Use (Beta) 🖉 Total Deformation Information Information Details of "Analysis Settings' Status Done Status Done MAPDL Elapsed Time 1 m 22 s Options 2 m 47 s MAPDL Elapsed Time Max Modes to Find MAPDL Memory Used 5.2793 GB 6 5.3281 GB MAPDL Memory Used ~40Mb N/A Limit Search to Range No MAPDL Result File Size MAPDL Result File Size 922.13 MB On Demand Expansion Yes Post Processing Post Processing Solver Controls Damped No Number of elements : ~420 000 Number of modes : 6 Solver Type Program Controlled + Rotordynamics Controls



 Reduce results file size and decrease overall time to solve by referencing Modal results files instead of copying them.

⊨ → 10 Harmonic Response 2 (C5)



Transient Structural

↓ ↓ □ ×

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× .

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Number of frequencies : 100

<u>10 Harmonic Response (B5)</u>



Modal & MSUP Elemental Temperature

 Element Temperatures is no longer being requested during solution to be stored in the Modal and Mode-superposition analysis results file, which reduces the result file size

Details of "Solution (B6)"	
+ Solution	
Adaptive Mesh Refinement	t
Max Refinement Loops	1.
Refinement Depth	2.
 Information 	
Status	Done
MAPDL Elapsed Time	1 m 22 s
MAPDL Memory Used	12.111 GB
MAPDL Result File Size	162.31 MB

D	etails of "Solution (B6)"	▼ ‡ □ ×
+	Solution	
-	Adaptive Mesh Refinement	
	Max Refinement Loops	1.
	Refinement Depth	2.
-	Information	
	Status	Done
	MAPDL Elapsed Time	1 m 15 s
	MAPDL Memory Used	12.111 GB
1	MAPDL Result File Size	119.69 MB



Reduced disk space with less data stored in results file

• Using On Demand Expansion, by default mode shapes are no more stored in the result file. Postprocessing is done extracting the data from the mode file(s) thus removing duplication of stored data:

Details of "Solution (B6)"		₽ □ ×
+ Solution		
Adaptive Mesh Refinement	:	
Max Refinement Loops	1.	
Refinement Depth	2.	
Information	2	
Status	Done	
	10.111 CP	
MAPDL Memory Used	12.111 GB	
MAPDE Result File Size	120.06 MB	


Multistage Cyclic Analysis

• Modal and Prestressed Modal analyses now support multiple harmonic indices definition. It allows to improve the accuracy of the results by enriching the solution.

Stage 1							
	Nodal Diameters						
HI	HI	N-HI	N+HI	2N-HI	2N+HI		
0	0	6	6	12	12		
1	1	5	7	11	13		
2	2	4	8	10	14		
3	3	3	9	9	15		

Stage 2					
	Nodal Diameters				
HI	HI	N-HI	N+HI	2N-HI	2N+HI
0	0	12	12	24	24
1	1	11	13	23	25
2	2	10	14	22	26
3	3	9	15	21	27
4	4	8	16	20	28
5	5	7	17	19	29
6	6	6	18	18	30





Submit linked analysis job without waiting for completion

 This feature enables the submission of linked analyses on DCS without needing to wait for the upstream system's completion, thus allowing users to close Mechanical and conserve license for other purposes.





Limitations

- The synchronous submission process is supported for most cases except if an analysis includes any of the following features:
 - Spot Welds
 - Bolt Pretension
 - Weak Springs
 - EM Transducer loading condition
 - Command (APDL) objects
 - Contact Splitting
 - Linking though imported loads

These features may require creation of additional elements or nodes during the solution process or the results to be downloaded and hence, Mechanical would need to wait for the completion of the upstream job on DCS.



New Feature:

Exposure of Hybrid Parallel option to run solutions in Mechanical

Value Provided:

- Reduced memory usage
- Effective use of hardware resource
- Improved scalability of large models with high load balance ratios



Why Hybrid Parallel?

- Reduced Memory usage : Hybrid parallel reduces memory usage by using less MPI processes per compute node compared to distributed memory parallel (DMP)
- Effective use of hardware resource: To use fixed amount of memory with DMP, the cluster must use a fewer number of processes per compute node to increase memory and solution efficiency. Hybrid parallel can be used to address this issue and the cluster resource will be utilized in its full capacity
- Improved scalability of large models with high load balance ratios: Hybrid parallel can improve efficiency and scalability of large models where the load balance ratio is not optimum and such examples may include the cases where the contact pairs cannot be split, or re-mesh happens in certain regions due to mesh adaptivity (Nonlinear Adaptive Region supported in Mechanical)



Hybrid Parallel exposure in Mechanical

- Hybrid Parallel solution is supported inside Mechanical for MAPDL solution. To enable it, the user need to go to Solve Process Settings and Advanced Properties option.
- Once enabled, the user can specify the number of Threads per process and based on the cores and number of threads per process, the number of processes required will be displayed. The number of processes will be cores divided by number of threads per process

🚰 Advanced Properties	\times
✓ Distribute Solution (if possible)	
Max number of utilized cores: 8	
Hybrid Parallel (Mechanical APDL)	
Threads per process: 2	
Number of processes: 4	
Use GPU acceleration (if possible) None	
Number of utilized GPU devices: 1	
Manually specify Mechanical APDL solver memory settings	
Workspace: 0 MB	
Database: 0 MB	
Additional Command Line Arguments:	
OK Cancel	



Hybrid Parallel exposure in Mechanical

 Once the solution is executed in Hybrid parallel mode, the solution statistics will display that the solution was done using the Distributed and Shared Memory parallel option and it will also show the cores, processes and threads per process used for the simulation





Hybrid Parallel limitation in Mechanical

• The solver can automatically switch to Hybrid parallel mode for some cases where the distributed solution is turned on and Hybrid parallel mode is not enabled inside Mechanical.

This happens for cases where solver analyze that Hybrid parallel mode is efficient than Distributed parallel. In this mode, there is limitation that user cannot run the downstream linked analysis.

Hence, for this rare scenarios, it is recommended to explicitly turn on the Hybrid parallel option in Mechanical and re-run the upstream analysis before proceeding for downstream analysis.

For example, for pre-stress full harmonic analysis, if the upstream static analysis has automatically enabled the hybrid parallel mode, then the downstream harmonic analysis may not run successfully in Mechanical



New Feature:

Coupled Field analysis exposure in Mechanical

Value Provided:

- Enables easier workflows for sensor design, MEMs devices and actuators
- Ability to model piezoelectric and acoustic degrees of freedom
- Includes a library of commonly used piezoelectric materials used in sensor design



Piezoelectric coupling in Coupled Field Transient

Structural and Electric (Charge based) physics interaction through Piezoelectric coupling is supported for Coupled Field Transient Analysis. The physics region can be inserted to apply this physics interaction. Structural and Electric Boundary conditions are available for the analysis

-	Scope					
	Scoping Method	Geometry Selection				
	Geometry All Bodies					
-	Definition					
	Structural	tructural Yes				
	Acoustics	No				
	Thermal	No				
	Electric Charge					
	Suppressed	No				
-	Coupling Options					
	Piezoelectric	On				





Piezoelectric coupling in Coupled Field Transient

- Structural and Electric results and probes are supported
- The user can also plot the Charge convergence under Solution output
- As shown in the example, Voltage is being generated as a results of applied mechanical stress due to Direct piezoelectric effect









Piezoelectric-Acoustics coupling in Coupled Field Transient

- Piezoelectric coupling along with interaction of Acoustics physics using the Fluid Solid interface can be performed in Coupled Field Transient analysis
- Physics region object is used to define bodies with Acoustics, Structural or Structural-Electric (Charge based) physics. The boundary conditions associated to Acoustics, Structural and Electric physics can be defined in the analysis

	etalis of Physics i	Region				
-	Scope					
	Scoping Method	Named Selection				
	Named Selection piezo_body					
1	Definition					
	Structural	Yes				
	Acoustics	No				
	Thermal	No				
	Electric	Charge				
	Suppressed	No				
]	Coupling Options					
	Piezoelectric	On				

l	Scope		
	Scoping Method	Named Selection	
	Named Selection	fluid_body	
1	Definition		
	Structural	No	
	Acoustics	Yes	
	Thermal	No	
	Electric	No	
	Suppressed	No	
3	Acoustic Domain Definition		
	Artificially Matched Layers	Off	
3	Advanced Settings		
	Reference Pressure	2.e-005 Pa	
	Reference Static Pressure	1.0133e+005 Pa	

Compressible

Fluid Behavior





Piezoelectric-Acoustics coupling in Coupled Field Transient

- Acoustics, Structural and Electric results and probes are available. Plots of Charge convergence, Heat convergence, Force convergence can be seen on Solution output
- For example, the Acoustic pressure distribution can be seen on Acoustic body. Energy probe is also shown.





Ansys

Sample piezoelectric material properties in Engineering Data

- A library of common piezoelectric materials is provided in Engineering Data Sources
- These can be readily added to the project for the analysis and updated with material supplier provided data

Engineering Data Sources				→ ↓ X	
AB	С		D	^	
1 Data Source 🥖	Location		Descrip	tion	
4 III General Materials		General use material samples for use in var	rious analys	es.	
5 🗰 Additive Manufacturing Materials		Additive manufacturing material camples fr	or use in adr	litive manufacturing analyses	
6 🎒 Geomechanical Materials	Outline of Piezoel	ectric Materials			-
7 🗰 Composite Materials		A	B C	D	E
8 🎬 General Non-linear Materials	1	Contents of Piezoelectric Materials	Add	Source	Description
9 🎬 Explicit Materials	2 ⊟ Mat	erial			
10 🎬 Hyperelastic Materials		P. Preise Threads Crushel (Pr.T.O.2)	-	Prince ale atrice Materiale and	Berlincourt D., Jaffe H. (1958) Elastic and piezoelectric
11 III Magnetic B-H Curves	_ 3	Barium Intanate Crystal (BariOS)	5	Piezoelectric_Materials.xmi	111. pp. 143-148
12 III Thermal Materials	4	📎 Lithium Niobate (LiNbO3)	4	Piezoelectric_Materials.xml	Tiersten H.F "Linear Piezoelectric Plate Vibrations," Plenum Press, New York (1969)
14 III Piezoelectric Materials	5	📎 Lithium Tantalate (LiTaO3)	æ	Piezoelectric_Materials.xml	Tiersten H.F "Linear Piezoelectric Plate Vibrations," Plenum Press, New York (1969)
	6	📎 Lithium Teraborate (Li2B4O7)		Piezoelectric_Materials.xml	V. Petrov,R. Komatsu and T. Sugawara. Temperature tuned noncritical phase-matchingin Li2B407 for generation of cw laser radiation at 244 nm. Electron.Lett. 35 (1999) 721-2
	7	♥ PZT-26	4	Piezoelectric_Materials.xml	C. Bricault, C. Pezerat, M. Collet, A. Pyskir, P. Perrard, et al. . Multimodal reduction of acoustic radiation of thin plates by using single piezoelectric patch with a negative capacitance shunt. Applied Acoustics, Elsevier, 2019,145,pp.320-327
	8	№ PZT-4	4	Piezoelectric_Materials.xml	J Yang, Analysis of piezoelectric Devices (Appendix II), World Scientific Publication, Hackensack N.J. ISBN 9789812568618
	9	₲ PZT-5H	4	Piezoelectric_Materials.xml	H. A. Kunkel, S. Locke and B. Pikeroen, "Finite-element analysis of vibrational modes in piezoelectric ceramic disks," in IEEE Transactions on Ultrasonics, Ferroelectrics, and Frequency Control, vol. 37, no. 4, pp. 316-328, July 1990, doi: 10.1109/58.56492.
	10	📎 PZT-8	4	Piezoelectric_Materials.xml	J Yang, Analysis of piezoelectric Devices (Appendix II), World Scientific Publication, Hackensack N.J. ISBN 9789812568618
	11	📎 Quartz (alpha)	4	Piezoelectric_Materials.xml	Basic Material Quartz and Related Innovations. In: Piezoelectricity. Springer Series in Materials Science, vol 114. Springer, Berlin, Heidelberg



ThermoElectric coupling in Coupled Field Static analysis

- A Thermal Electric coupling can be performed in Coupled Field Static analysis.
- The analysis can account for thermoelectric effects, such as Joule heat, Seebeck, Peltier and Thomson effect. The coupling is included either at Load vector level or Matrix level
- The physics region can be specified with physics as Thermal only, Electric Conduction only or both. Thermal and Electric boundary conditions can be defined

D	Details of "Physics Region" 👻 🖣 🗖 🗙					
-	Scope					
	Scoping Method	Geometry Selection				
	Geometry	All Bodies				
-	- Definition					
	Structural	No				
	Acoustics	No				
	Thermal	Yes				
	Electric	Conduction				
	Suppressed	No				





ThermoElectric coupling in Coupled Field Static analysis

- Thermal and Electric results and probes are available
- Heat Convergence and Current convergence plots can be seen on Solution Output.
- For example, the temperature and voltage distribution of a Thermoelectric cooler is shown below.





Structural-ThermoElectric coupling in Coupled Field Static analysis

- Coupling of Structural and Thermoelectric physics can be performed in Coupled Field Static analysis. The coupling effects of both Structural-Thermal using thermal strain and Thermal-Electric conduction based of Joule heating, Seebeck, Peltier and Thomson effects will be considered when Structural-Thermal-Electric Conduction based physics is specified using Physics region.
- Structural, Thermal and Electric boundary conditions can be defined in the analysis

D	etails of "Physics I	Region" 🔻 🕂 🗖 🗙				
-	Scope					
	Scoping Method	hod Geometry Selection				
	Geometry	1 Body				
-	Definition					
	Structural	Yes				
	Acoustics	No				
	Thermal	Yes				
	Electric	Conduction				
	Suppressed No					
3	Coupling Option:					
	Thermal Strain Program Controlled					

D	etails of "Physics I	Region 2" 🔻 🕇 🗖 🗙				
-	Scope					
	Scoping Method	Geometry Selection				
	Geometry	1 Body				
Ξ	Definition					
	Structural	No				
	Acoustics	No				
ſ	Thermal	Yes				
	Electric	Conduction				
	Suppressed	No				





Structural-ThermoElectric coupling in Coupled Field Static analysis

- Structural, Thermal and Electric results and probes are available. The user can plot Force Convergence, Heat Convergence and Current convergence plots from Solution output.
- The animation of the deformation and voltage of an Electro Thermal Micro actuator is shown below





Linear Periodic symmetry for Voltage (Coupled Field Analysis)

- In previous releases linear periodic objects can be scoped to only structural or thermal-electric bodies. From 2022 R2, Linear periodic type symmetry region can be applied in Coupled Field Analysis with Displacement and Voltage degree of freedom.
- Use **Apply To** property to specify the Degree Of Freedom(DOF):
 - Applicable DOF (Default)
 - specifies Displacement and Voltage DOFs on the geometries of the Low Boundary and High Boundary.
 - Displacement
 - specifies Displacement DOF on the geometries of the Low Boundary and High Boundary.
 - Voltage
 - specifies Voltage DOF on the geometries of the Low Boundary and High Boundary.

=	Scope					
	Scoping Method	Geometry Selection				
	Low Boundary	1 Face				
	High Boundary	1 Face				
-	Definition					
	Scope Mode	Manual				
	Туре	Linear Periodic				
	Behavior	Free				
	Apply To	Applicable DOF				
		Applicable DOF Displacement				



Summary of Coupled Field systems

	Coupled Field Harmonic	Coupled Field Modal	Coupled Field Static	Coupled Field Transient
Structural- Thermal	[1]	[2]	Thermal strain Thermoplasticity Thermoviscoelasticity	Thermal strain Thermoelastic Damping Thermoplasticity Thermoviscoelasticity
Structural - Acoustics	Fluid Solid Interface	Fluid Solid Interface	Fluid Solid Interface	Fluid Solid Interface
Structural Electric (Charge)	Piezoelectric	Piezoelectric	Piezoelectric	Piezoelectric 🔶 📩
Structural Electric Acoustic	Piezoelectric Fluid Solid Interface	Piezoelectric Fluid Solid Interface	Piezoelectric Fluid Solid Interface	Piezoelectric \bigstar Fluid Solid Interface
Thermal Electric (Conduction)	[2]	[2]	Joule Heating Seebeck effect Peltier effect Thomson effect	[1]
Structural Thermal Electric (Conduction)	[2]	[2]	Joule Heating Seebeck effect Peltier effect Thomson effect Thermal strain	[1]

1 Not Supported in Mechanical

2 Not Applicable



Transient piezoelectric and piezo-acoustic

- Piezoelectric haptic feedback devices
- Pulse based piezoelectric transducer design
- Acoustic imaging/NDE through piezoelectric transmitter/receiver
- Piezoelectric micro-speaker design
- Many others

Static structural-thermoelectric

- Joule heating in various applications
- Electric connectors, splice design
- Thermocouple design
- Thermoelectric cooler and generator design
- Thermoelectric Micro-actuators
- Many others



New Feature:

Hyper viscoelastic materials in Linear Perturbation full harmonic analysis

Value Provided:

• Allows for the harmonic analysis of preloaded hyper viscoelastic materials.



Hyper Viscoelasticity in Perturbed Harmonic Analyses

4000

3500

3000

2500

2000

1500

1000

500

- Harmonic analysis is important in areas like vibration isolation and noise transmission
- For Noise, Vibration and Harshness (NVH) in vehicle design, the structures include plastics, foams, and other vibration absorbing materials
 - It is important to capture the deformed and prestressed state of the material during harmonic analysis
- This new capability will allow users to simulate the deformation in a static or transient step and then perform a followon perturbed harmonic analysis to determine the vibration characteristics of the structures that have hyper viscoelastic materials

Vibration Isolation Mount, Yeoh hyperelastic with Prony series viscoelasticity, compressed 30% and experiencing a fixedamplitude small vibration

Real(F)

Imaginary(F)

Force Transmission vs. Vibration Frequency shows the damping behavior (Imaginary part) is greatest at about 0.6 Hz



New Feature:

Support for AMD GPUs

Value Provided:

• Faster simulations (HPC value)



- Added support for Instinct GPUs from AMD
 - New generation (CDNA 2.0) cards significantly faster than previous generation (CDNA 1.0)
 - Uses HIP/ROCm 5.0 **7** requires AMD driver version 21.50 or newer

AMDA ROCM





- Added support for Instinct GPUs from AMD
 - Focused on sparse direct solver
 - Targeting Linux only this release

Model	Release Date	Memory Capacity (GB)	Memory Bandwidth (GB/s)	Peak FP64 Compute (TFlops)	Peak FP32 Compute (TFlops)	Peak TF32 Compute (TFlops)
AMD Instinct MI100	November 2020	32	1223	11.5	23.1	N/A
NVIDIA A100	May 2020	40/80	1555/1935	9.7 (19.5 ¹)	19.5	156
AMD Instinct MI210	December 2021	64	1638	22.6	22.6	N/A
AMD Instinct MI250	November 2021	128	3277	45.3	45.3	N/A

¹ Compute speed when using tensor cores

• Data provided by https://en.wikipedia.org



• Significantly faster performance for direct solver benchmarks





• Significantly faster performance for direct solver benchmarks





• Significantly faster performance (Engine block model)





New Feature:

Default number of cores increased from 2 to 4

Value Provided:

- Better HPC value
- When the product launches by default using only 2 CPU cores, it is underutilizing what they already paid for



Distributed Memory Parallel Enhancements

- Default changed from 2 to 4 CPU cores
- Replaced "Distributed ANSYS" naming "Distributed Memory Parallel"
- MPI library support
 - Upgraded to Intel MPI 2021 Update 6 on Windows and Linux
 - Improves performance, scalability and robustness
 - Linux clusters using (older) Mellanox Infiniband 4.x ⑦ (older) Intel MPI 2018 is automatically chosen
 - Microsoft MPI v10.0 is unchanged at this release on Windows
 - Open MPI v4.0.5 is unchanged at this release on Linux



Distributed Memory Parallel Enhancements

• Improved scaling at higher core counts



50 MDOF; Sparse direct solver
Nonlinear, static analysis involving large deflections, 20 equilibrium iterations, bonded contact pairs
Linux cluster; each compute node contains 2 Intel Xeon Platinum 8260L processors (48 cores), 192GB RAM, SSD, CentOS 7.7, Mellanox HDR Infiniband





New Feature:

SMART enhancements: Automatic Crack Initiations and Remeshing Mesh Sizing Control

Value Provided:

- Broaden simulation scopes
- Enable users to better control meshes
- Improve solution accuracy and robustness



SMART Enhancement: Multi Cracks Initiation

- Multiple crack initiations
 - Multiple cracks can be initiated simultaneously or sequentially
- Maximum principal stress based initiation criteria
- Sizing control of initiated crack
- Support both fatigue and static crack growth











SMART Enhancement: Automatic Crack Initiation

- Automatic crack initiation with preexisting crack(s)
 - Automatic crack initiation supports now structures with pre-existing crack(s)
- Sizing control of initiated crack
- Maximum principal stress-based initiation criteria
- Support both fatigue and static crack growth

Support multiple pre-existing cracks and multiple crack initiations



Stress solution of initial model

After crack-initiation





SMART Enhancement: Mesh Sizing Control

- SMART remeshing mesh sizing control
 - Sizing control with specific node or element components
 - Sizing control with specific locations (coordinates)



Mesh sizing control to maintain geometry

Fatigue crack growth Mesh coarsening option: CONS





SMART Enhancement: Mesh Sizing Control

• Perforated plate with two edge cracks

Fatigue crack growth Mesh coarsening option: AGGR



Mesh Oracle

Without mesh sizing control

With mesh sizing control




New Feature:

Composite post-processing functionalities moved into DPF plugin

Value Provided:

- More efficient post-processing
- More efficient and robust
- Easy to expose the composite post-processing functionalities in different contexts such as pyAnsys
- Flexibility for the future



ACP – LS-DYNA Workflow in Workbench

- Analyze composite structures with respect to crash and impact
- Supports shell models as *ELEMENT_SHELL, solid models as *ELEMENT_TSHELL (beta) and any kind of assemblies (new)
- Many LS-DYNA material cards are now available in **Engineering Data** allowing the definition of a material card per ply material



74

Composite Post-Processing in Mechanical

- Composite post-processing is a new plugin of DPF
- Used by the existing Composite Failure Tool and Sampling Point Tool in Mechanical
- More efficient and more robust
- Allows for more flexible post-processing in the future (pyAnsys)



Transferring the Lay-up on a Solid Mesh

- The mapping can now be restricted to sub-domains of the solid mesh.
- With the new Lay-up Mapping Objects it is possible to define exactly which plies are transferred onto which regions (Mesh Components) of the volume mesh.
- This improves the handling of complex and even hybrid structures





Lay-up Mapping: Modeling a T-Joint *

1. Create a solid mesh with Named Selections



2. Pass the solid mesh (C4) to ACP Pre (D5)



3. Define lay-up mapping: source (plies) and target (Mesh Components)







Section Cut Scoping

- Section Cuts allow visualizing and exporting the composite lay-up definition on an arbitrary section plane
- An option was added to restrict the scope of a Section Cut to selected Element Sets
- This makes Section Cuts easier to interpret, especially in complex geometries



Section Cut with ply-wise angles



L Section	on Cut Prope	erties		_			
Name:	SectionCut.1						
ID: S	ectionCut.1						
General	Wire Frame	Options	Surface Optio	ns			
Active:							
Positio	n						
Intera	ctive Plane:	\checkmark					
	Origin:	(83.2290	, 2.5000, -0.0000)			
	Normal:	(0.0000,	0.0000, 1.0000)				
Refer	ence Directio	n 1: (1.0	000, 0.0000, 0.00	000)	Flip		
Show	Plane: 🗹						
Scopin	g						
Entire	Model:						
Eleme	nt Sets: ['el	s_huelle', '	'els_gurt']				
Extrusio	on						
	Type:	Wire Fran	ne		~		
9	Scale Factor: 0.000 🦛 📕 🖨 🖨 11.000 1.000						
Core Scale Factor: 1.0							
Sectio	on Cut Type:	Modeling	g Ply Wise		~		
			ОК	Apply	Cancel		



Graphical User Interface

- The UI of ACP has been improved in these areas
 - Use **F2** for renaming objects directly via the Object Tree.
 - Hot keys were added for Copy & Paste (Ctrl + C, Ctrl + V). Multi-selection is also supported.
 - The Object Tree now always shows automatically the latest state (up-to-date, out-of-date) of the objects.



 Vector plots such as fiber directions or orientations were further refined. Use the Plot Properties dialog from the toolbar to manually adjust the settings (scaling factor and density).



🚼 Plot Properties		_		×
Vector Scaling Fa Reduce Number of Displayed Vectors by Fa	ctor: 0.000 🌾 🚃 🖡		20.000 10	.000
Reset to Defaults	Set Values to 1 Set Value	s to 5	Set Values	to 10
		ОК	Cano	cel





New Feature:

Multizone Hex Meshing

Value Provided:

- Obtain hex elements with less requirements with CartSweep and ThinSweep decomposition
- Improved robustness of automatic-thickness



Thin Sweep Decomposition for Multizone

No
MultiZone
Thin Sweep

- **Thin Sweep** is a New Multizone Decomposition option for thin bodies
- Blocking technology is used to create and mesh 2D block on one side and pull automatically into 3D block
- In 22R2 we begin to support some simple bodies where connectivity is required
 - Share topology connections from two sides working on simple cases





Cart Sweep Decomposition for Multizone

- **Cart Sweep** is a New Multizone Decomposition option for 2.5D bodies
 - Sweepable in one Cartesian direction
 - Direction can be specified by picking source faces
- A Cartesian background grid is used to create blocks
- Allows users to mesh with hex elements with less requirement to decompose into hex-able volumes in geometry tools

Definition

Suppressed	No
Method	MultiZone
Decomposition Type	Cart Sweep











New Feature:

Enhanced Material Support for the LS-DYNA Solver

Value Provided:

- Simplifies the modeling process
- Enables Advanced Material Modeling for a wide range of applications including Drop Test, Impact, and Composites Application



Material Improvements



Improved Material Support

- Additional fields are made available for the definition of LS-DYNA specific material MAT_SIMPLIFIED_RUBBER/FOAM
- Materials added:
 - MAT_OGDEN_RUBBER including all input card variations and optional cards
 - MAT_GENERAL_VISCOELASTIC

	A	В	с	D	E
1	Property	Value	Unit	8	ſŗ,
2	*MAT_GENERAL_VISCOELASTIC				
3	Definition				
4	Tensile Pressure Elimination Flag, pcf	0			
5	Elastic Flag, ef	0			
6	Number of Terms in Shear Fit, nt	0			
7	Shear Fit Coefficient, bstart	0	s^-1	· I	
8	Optional Ramp Time for Shear Loading, tramp	0	s	· I	
9	Number of Terms in Bulk Fit, ntk	0			
10	Bulk Fit Coefficient, bstartk	0	s^-1	·	
11	Optional Ramp Time for Bulk Loading, trampk	0	s	·	
12	Initial Shear Modulus, G0	1	Pa	·	
13	Initial Bulk Modulus, KO	1	Pa	·	

	SDYNA External Model - MAT
12	*MAT_ELASTIC
1	*MAT_ORTHOTROPIC_ELASTIC
1	*MAT_ANISOTROPIC_ELASTIC
1	*MAT_PLASTIC_KINEMATIC
7	*MAT_BLATZ-KO_RUBBER
1	*MAT_HIGH_EXPLOSIVE_BURN
1	*MAT_NULL
7	*MAT_JOHNSON_COOK
1	*MAT_POWER_LAW_PLASTICITY
7	*MAT_PIECEWISE_LINEAR_PLASTICITY
7	*MAT_ENHANCED_COMPOSITE_DAMAGE
7	*MAT_LAMINATED_COMPOSITE_FABRIC
1	*MAT_CRUSHABLE_FOAM
P	*MAT_OGDEN_RUBBER
7	*MAT_SIMPLIFIED_JOHNSON_COOK
1	*MAT_MODIFIED_PIECEWISE_LINEAR_PLASTICITY
7	*MAT_SIMPLIFIED_RUBBER/FOAM
17	*MAT_BILKHU/DUBOIS_FOAM
1	*MAT_FABRIC
1	*MAT_ADD_EROSION
12	*MAT_ADD_DAMAGE_GISSMO
7	*MAT_GENERAL_VISCOELASTIC



Improved Material Support

Many of the constitutive models in LSDYNA do not incorporate failure and erosion in the main model.

The following Material Models additions provide a way of including damage and failure in these models.

*MAT_ADD_EROSION *MAT_ADD_DAMAGE_GISSMO

Project 🔗 A	2:Engineeri	ng Data 🗙								
Y Filter Engineering Data III Engineering D	ata Sources	Reload Extensions								
Toolbox · 부 : X'di Dringipal Stragg Failura	× Outline	of Schematic A2: Engineering Data					And the second se		¥	д >
Principal Strain Failure	·	A	В	С	D		E			
Principal Strain Failure	1	Contents of Engineering Data	Source	Description						
Cal Johnson Cook Failure	2	🗖 Material								
E Forming Plasticity	3	So Material With Failure	T	m						
2 Bilinear Transversely Anisotropic Harr	*	Click here to add a new material		line i						_
Multilinear Transversely Anisotropic H				_						
2 Bilinear ELD Transversely Anisotropic										
Multilinear FLD Transversely Anisotropic										
Bilinear 3 Parameter Barlat Hardening										
Exponential 3 Parameter Barlat Harder										
Exponential Barlat Anisotropic Hardeni										
E Foams										
Rate Independent Low Density Foam										
El Eulerian										
Ma Vacuum										
Cal Concrete EC2 (Beta)										
SDVNA External Model - MAT										
Cal	Description		_	_						
	Propert	des of Outline Row 4: Material With Failure								4 7
2 *MAT ANISOTROPIC FLASTIC		A				В	С		D	E
MAT PLASTIC KINEMATIC	1	Property				Value	Unit		8	(p2
MAT_BLATZ-KO_RUBBER	2	🔁 Material Field Variables				Table				
MAT_HIGH_EXPLOSIVE_BURN	3	🔁 Density				7800	kg m^-3	-		
MAT_NULL	4	Isotropic Elasticity						6		
MAT_JOHNSON_COOK	5	Derive from				Young's Modulus and				
MAT_POWER_LAW_PLASTICITY	6	Young's Modulus				1E+10	Pa	-		E
*MAT_PIECEWISE_LINEAR_PLASTICI	7	Poisson's Ratio				0.3			1	Ē
*MAT_ENHANCED_COMPOSITE_DAM	8	Bulk Modulus				8.3333E+09	Pa			F
*MAT_LAMINATED_COMPOSITE_FAB	9	Shear Modulus				3.8462E+09	Pa			F
2 MAT_CRUSHABLE_FOAM	10								m	-
TAL_UGDEN_RUBBER	10	Avial Dameira Easter da				0				
*MAT_MODIFIED_PIECEWISE INFAL	11	Panding Damping Factor, da				0				
S *MAT SIMPLIFIED RUBBER/FOAM	12								-	
A *MAT BILKHU/DUBOIS FOAM	13	E MAT_ADD_EROSION								
MAT FABRIC	14							_		
A MAT ADD EROSION	15	Time Period, dtefit				0	s	-		
*MAT_ADD_DAMAGE_GISSMO	16	Minimum Time Step Size at Failure, dtmin				0	S	-		
MAT_GENERAL_VISCOELASTIC	17	Maximum Effective Strain at Failure, effeps				0	m m^-1	•		
MAT_COMPOSITE_FAILURE_SHELL_	18	Critical Energy for Nonlocal Failure Criterion, engcrt				0	3	•		
*MAT_COMPOSITE_FAILURE_SOLID_	19	Tensorial Shear Strain at Failure, epssh				0	m m^-1	•		
*MAT_ORTHOTROPIC_SIMPLIFIED_D	20	Thinning Strain at Failure for thin and Thick Shells, epsth	in			0	m m^-1	-		
LSDYNA External Model - EOS	21	Exclusion Number, excl				0				
*EOS_LINEAR_POLYNOMIAL	22	Failure time, failtm				0	S		1	
EOS_JWL	23	Flag for Damage Model, idam				0				
EUS_GRUNEISEN	24	Stress Impulse for Failure, impulse				0	Pa	-		
	25	Minimum Principal Strain at Failure, mneps				0	m m^-1	*		
Custon Matadal Madda	26	Minimum Pressure at Failure, mnpres				0	Pa	-		
E Custom Material Models	27	Variable to Invoke a Failure Criterion Based on Maximum	Princi	ipal S	train, mxeps	0	m m^-1	-		
Create Custom Model	28	Maximum Pressure at Failure, mxpres			en mesterne feitiges	0	Pa	-		
								- Contraction of the local division of the l	_	



Improved Material Support

•Materials models targeting composite and turbomachinery applications have been

Added. They enable advanced ply modelling and failure

- *MAT_ENHANCED_COMPOSITE_DAMAGE
- *MAT_LAMINATED_COMPOOSITE_FABRIC
- *MAT_COMPOSITE_FAILURE_SHELL_MODEL
- *MAT_COMPOSITE_FAILURE_SOLID_MODEL
- *MAT_ORTHOTROPIC_SIMPLIFIED_DAMAGE

1	_SDYN/	A External Model - MAT	
2	*MAT	ELASTIC	
7	*MAT	ORTHOTROPIC ELASTIC	
2	*MAT	ANISOTROPIC ELASTIC	
2	*MAT	PLASTIC KINEMATIC	
2	*MAT	BLATZ-KO RUBBER	
2	*MAT	HIGH EXPLOSIVE BURN	
2	*MAT	NULL	
2	*MAT	JOHNSON COOK	
~	*MAT	POWER LAW PLASTICITY	
2	*MAT	DIECEWISE LINEAR DIASTICITY	
2	*MAT	ENHANCED COMPOSITE DAMAGE	
	*MAT	LAMINATED COMPOSITE FABRIC	
	*MAT	CRUSHABLE FOAM OGDEN RUBBER	
	*MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK	
	*MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY	
NNNN	*MAT *MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM	
NNNNNN	*MAT *MAT *MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM	
NNNNNNN	*MAT *MAT *MAT *MAT *MAT *MAT	CROSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC	
	*MAT *MAT *MAT *MAT *MAT *MAT *MAT	CROSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC ADD EROSION	
	*MAT *MAT *MAT *MAT *MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC ADD EROSION ADD DAMAGE GISSMO CENTERAL VISCOLLACTIC	
NNNNNNNNNN	*MAT *MAT *MAT *MAT *MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC ADD EROSION ADD DAMAGE GISSMO <u>CENERAL VISCOELASTIC</u> COMPOSITE FAILURE CUEL MODEL	
NNNNNNNNN	*MAT *MAT *MAT *MAT *MAT *MAT *MAT *MAT	CRUSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC ADD EROSION ADD DAMAGE GISSMO <u>CENERAL VISCOELASTIC</u> COMPOSITE FAILURE SHELL MODEL COMPOSITE FAILURE SHELL MODEL	
NNN NNNNNNNN	*MAT *MAT *MAT *MAT *MAT *MAT *MAT *MAT	CROSHABLE FOAM OGDEN RUBBER SIMPLIFIED JOHNSON COOK MODIFIED PIECEWISE LINEAR PLASTICITY SIMPLIFIED RUBBER/FOAM BILKHU/DUBOIS FOAM FABRIC ADD EROSION ADD DAMAGE GISSMO <u>CENERAL VISCOELASTIC</u> COMPOSITE FAILURE SHELL MODEL COMPOSITE FAILURE SOLID MODEL COMPOSITE FAILURE SOLID MODEL	



ALE Improvements



Improved Support for ALE

- Field added to specify reference pressure under analysis settings
- Option to use newer coupling algorithm (ALE_Structured_FSI) for S-ALE
- Ability to define coupling stiffness using tabular data

Lagrange Bodies Scoping Method Geometry Selection Geometry 1 Body ALE Bodies Scoping Method Geometry 1 Body ALE Bodies 1 Body Scoping Method Geometry Selection Geometry 1 Body Definition Fluid Structure Interaction Type Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	etails of "Coupling"	▼ ‡ 🗆 ×					
Scoping Method Geometry Selection Geometry 1 Body ALE Bodies Scoping Method Geometry 1 Body Definition 1 Body Fluid Structure Interaction Type Program Controlled Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Lagrange Bodies						
Geometry 1 Body ALE Bodies Scoping Method Scoping Method Geometry Selection Geometry 1 Body Definition Fluid Structure Interaction Type Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Scoping Method	Geometry Selection					
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Scoping Method Geometry Selection Geometry 1 Body Definition Fluid Structure Interaction Type Program Controlled Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	ALE Bodies						
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Definition Fluid Structure Interaction Type Program Controlled Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Geometry	1 Body					
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Fluid Structure Coupling Method Program Controlled Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Fluid Structure Interaction Type	Program Controlled 🔹					
Coupling Direction Constrained Lagrange in Solid ALE Structured FSI Number of Coupling Points 2 Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Fluid Structure Coupling Method	Program Controlled					
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Lagrange Normals Point Toward ALE Fluids Yes Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Number of Coupling Points	2					
Leakage Control None Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Lagrange Normals Point Toward ALE Fluids	Yes					
Stiffness Type Tabular Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Leakage Control	None					
Stiffness Load Curve Tabular Data Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Stiffness Type	Tabular					
Minimum Volume Fraction to Activate Coupling 0.5 Friction 0 Birth Time 0 s Death Time 1E+20 s	Stiffness Load Curve	Tabular Data					
Friction 0 Birth Time 0 s Death Time 1E+20 s	Minimum Volume Fraction to Activate Coupling	0.5					
Birth Time 0 s Death Time 1E+20 s	Friction	0					
Death Time 1E+20 s	Birth Time	0 s					
	Death Time	1E+20 s					

ALE Controls	
Continuum Treatment	Use Alternate Advection Logic
Cycles Between Advection	1
Advection Method	Donor Cell + Half Index Shift
Simple Average Weighting Factor	-1
Volume Weighting Factor	0
Isoparametric Weighting Factor	0
Equipotential Weighting Factor	0
Equilibrium Weighting Factor	0
Advection Factor	0
Start	0 s
End	1E+20 s
Reference Pressure	0 Pa
Advanced	
Output Controls	





Structured FSI Coupling Method

The new structured FSI coupling method can now be used for S-ALE simulations

Details of "Coupling"	▼ ‡ □ ×					
Lagrange Bodies						
Scoping Method	Geometry Selection					
Geometry						
ALE Bodies						
Scoping Method	Geometry Selection					
Geometry						
Definition						
Fluid Structure Interaction Type	Program Controlled 🔹					
Fluid Structure Coupling Method	Program Controlled					
Coupling Direction	ALE Stuctured FSI					
Number of Coupling Points	2					
Lagrange Normals Point Toward ALE Fluids	Yes					

1	*ALE_STRUCTURED_FSI										
1	\$ slave	master	sstyp	mstyp	unused	unused	unused	mcoup			
	1	45	1	1				-2			
1	\$ start	end	pfac	unused	unused	flip	unused	unused			
	0	1E+20	0.1			0					



SPH Improvements



Adaptive Solid to SPH (Material Assignment)

- SPH particles existing inside the solid element are inactive initially and become active when the solid element reaches failure criteria defined by the material of the solid part.
 - The SPH particles are created inside the solver.

If the user want the generated particles (after failure) to have different material, he/she should set material Assignment to User Defined. In the field Material, we can access all materials defined in the model.



Details of "Adaptive Solid To SPH"	·····································				
Geometry					
Scoping Method	Geometry Selection				
Geometry	1 Body				
Definition					
Per Element Direction number of particles	1				
Coupling Type	Debris				
Coupling Start	From Beginning				
Material Assignment	User Defined				
Material	Structural Steel 🔹				
Section SPH controls					
Smoothing Length Constant	1.2				
Maximum Scale Factor	2				
Minimum Scale Factor	0.2				



Adaptive Solid to SPH (Material Assignment)

- The Adaptive Solid To SPH particles are created inside the solver
- The SPH part is not visible on mechanical and cannot be scoped in contact.
- A child object under the corresponding adaptive solid to SPH called Contact SPH to Target Bodies can be added.
- The user should set the target Bodies and the contact properties.
- The user can run the contact algorithm in MPP configuration by setting the processing type to MPP in analysis settings
- The user can add erosion to contact by Eroding property to Yes

	⊡√ੴ Mesh □	ng) nditior Setting	is 35				
	Adaptive		Insert			•	1
	⊡ <mark>γ</mark> ⊡ Solution ⊕	* ₀	Suppress				
			Duplicate				
D	etails of "Adaptive Solid To		Сору				→ ┦ □ ×
Ξ	Geometry	~	Cut				
	Scoping Method		Cut				0
	Geometry	ها	Copy To (lipboa	ard		
Ξ	Definition	×	Delete				
	Per Element Direction num	Th	Dename		E2		
	Coupling Type	<u>a</u> ro	Rename		12		
	Coupling Start		Group		Ctrl+G		
	Material Assignment		Group Sir	nilar O	bjects		
	Material		Add Cont	act			•
-	Section SPH controls	_	Add Com	ucc			
	Smoothing Length Consta	nt		1.2			
	Maximum Scale Factor			2			
	Minimum Scale Factor			0.2			

Oetails of "Contact SPH to Target Bodies"	▼ ‡ □ ×			
Target Bodies				
Scoping Method	Geometry Selection			
Geometry				
Definition				
Formulation	ERODING_NODES_TO_SURFACE_MPP			
Eroding	Yes			
MPP	Yes 💌			
Sort Frequency	100			
Contact ID	179			
Common Controls				
Birth Time	0 s			
Death Time	0 s			
Viscous Damping Coefficient	10			
Contact Penalty Scale Factor	1E-12			
Target Penalty Scale Factor	1E-12			
Advanced Controls				
Optional Thickness for Contact Surface	0 m			
Optional Thickness for Target Surface	0 m			
Soft Constraint Formulation	Program Controlled			
Soft Constraint Scale Factor	0.1			
Depth	5			
Eroding Controls				
Symmetry Plane Option	Program Controlled			
Erosion Interior Node Option Program Controlled				
Solid Elements Treatment Program Controlled				



New Feature:

Fully automated element embedding workflow for modeling intersecting components of elements

Value Provided:

- New workflow greatly reduces the model preparation time and computational costs
- Integrated seamlessly with the released Reinforcing capability
- Powerful usability and accuracy enhancements by harvesting all embedded element features



Direct Element Embedding Workflow

- The connections between two intersecting elements or element components (embedded and base elements) can now be automatically established with a new Direct Embedding procedure. A new command **EEMBED** is introduced
- Structural beams/links and thermal links are supported for embedded members
- Extensive applications: electronics reliability, composites, biomedical







Advantages of Direct Element Embedding

- Can be seamlessly integrated with current Reinforcing workflow (see example on the right)
- Allows both embedded members and base materials to be modeled with standard structural or thermal elements
- Accounts for the actual cross-section geometry of the embedded members in embedded / base element connections.
- Improves the modeling accuracy (for instance, inclusion of bending and torsional stiffness for embedded members is now feasible using standard beam elements)
- Effectively eliminates the solution sensitivity with respect to the mesh density.



SOLID185 – Concrete Encasement Reinforcing - Rebars Direct embedding – Steel Section

New Feature:

Co-Simulation of Aqwa, Rigid Dynamics and AeroDyn for the offshore wind turbine

Value Provided:

Enables user to simulate the coupling system
 of hydrodynamics, mooring dynamics,
 aerodynamics, and structural dynamics of the
 offshore floating wind turbine, providing the
 transient responses and stresses on the
 floating wind turbine system under various sea
 and wind conditions



Floating Wind Turbine Co-simulation Workflow

 Provides an integrated procedure for the time domain hydrodynamics, mooring dynamics, aerodynamics and structural dynamics response analysis for floating wind turbine systems



Pre and Post Processing of Co-simulation in Workbench

• Installation:

- AqwaCosimulation Extension pre-installed with Workbench
- Manually loaded from Extension > Manage Extensions
- Inputs:
 - Model geometry and mooring definition
 - Ocean environment settings
 - Blade and tower sectional aerodynamic data
 - Generator torque vs. Speed curve
- Results:
 - Time history of responses
 - Flexible structural analysis





New Feature:

Structural Optimization

Value Provided:

• New optimization methods and constraints open up more possibilities for customers to create optimized designs.



New Features & Capabilities

- New manufacturing constraint
- Design constraint



Topology Optimization - Housing





- Topology Optimization delivers the lightest solution. However, designs are almost systematically perforated which makes them inappropriate for some contexts.
- For example, consider a housing that is clamped on another component and contains a liquid.
- Would Topology optimization be capable to deliver a watertight design?
- Where should the envelop be optimally located?













Topology Optimization - Housing



 $\begin{cases}
|\min_{\Omega} compliance| \\
|vol < 30\%
\end{cases}$

 $\begin{cases} \min_{\Omega} compliance \\ vol < 30\% \\ | + housing (3 faces to enclose) \end{cases}$



 $\begin{cases} \min_{\Omega} compliance \\ vol < 30\% \\ +housing (5 faces to enclose) \\ \lfloor +2\text{-sided z-dir pullout} \end{cases}$















• Given a working domain and a stamping direction, can topology optimization give any guidance to sketch plate-like design?



D	etails of "Manufacturing Const	raint 2" coordooccoordoo		oooooooooo 🗢 🕂 🗖 🗙	
-	Scope				
	Scoping Method	Optimization Regio	n	About the cot i	
	Optimization Region Selection	Optimization Regio	n	About the set-t	, h
-	Definition	1		1)create a pull-	out manuf.
	Туре	Manufacturing Con	straint	constraint	
	Subtype	Pull Out Direction		2) select the « s	tamning » ontion
	Suppressed	No			
	Pull Out Option	Stamping		3)define the pu	flout direction
-	Location and Orientation			corresponding t	to your stamping
	Coordinate System	Global Coordinate	System	direction	
	Axis	Z Axis			
	Direction	Both Directions			

$\begin{cases} |\min_{\Omega} compliance| \\ |vol < 40\% \end{cases}$

By nature, Topology Optimization is not capable to deliver plate design, it rather favors massive design.





No-hole (iso-topology)



- In casting process it may be requested to have the simplest design as possible in a bid to ease the filling process. Any perforation or hole is somehow an obstacle that makes the filling stage more delicate and energyconsuming.
- The so-called « no-hole » feature aims to fulfill this manufacturing constraint.

D	etails of "Manufacturing Constr	raint"				
- Scope						
	Scoping Method	Optimization Region				
	Optimization Region Selection	Optimization Region				
-	Definition					
	Туре	Manufacturing Constraint	Abou			
	Subtype	Pull Out Direction	ADOU			
	Suppressed	No	1)cre			
	Pull Out Option	No-Hole	const			
-	Location and Orientation					
	Coordinate System	Global Coordinate System	ontio			
	Axis	Y Axis	υριιο			
	Direction	Both Directions				

About the set-up			
1)create a pull-out manuf.			
constraint			
2)select the « no-hole »			
option			
•			



Known limitation: The no-hole constraint can be sometimes slighlty violated.



Topology Optimization

Design constraint with Level-Set

- Design constraints are now available in Level-set based Topology Optimization:
 - cyclic symetry,
 - plane symetry
 - and pattern repetition.

Aechanical setup	
with 3 loadsteps	



Mechanical setup with 2 loadsteps

 $\left(\min_{\Omega}\sum_{k}compliance_{k}\right)$ $mass \leq 35\%$ +4th y-cyclic +1-sided y-pullout

 $\left(\min_{\Omega}\sum_{k} complianc\right)$

 $mass \leq 35$

+4th







Details of "Design Constraint" 🗢 🖣 🗖 🗙					
Ξ	Scope				
	Scoping Method	Optimization Region			
	Optimization Region Selection	Optimization Region			
-	Definition				
	Туре	Design Constraint			
	Subtype	Cyclic Repetition			
	Suppressed	No			
-	Location and Orientation				
	Number of Sectors	4			
	Coordinate System	Global Coordinate System			
	Axis	X Axis			



 $\begin{cases} \min_{\Omega} \sum_{k} compliance_{k} \\ mass \leq 35\% \\ +4th y-cyclic \end{cases}$













Min Gap

- This new manufacturing constraint aims to keep a minimum distance between features. It is motivated by the casting process to increase the life cycle of the mold.
- This constraint is qualitative. The formulation is based on an approximation that aims to limit the amount of material within multiple test regions. This capability should be used in combination with max-thickness.

Recommandation : GapSize $\leq 2.maxThick$

Details of "Manufacturing Constraint" 👻 🗖 🗖 🗙						
-	Scope					
	Scoping Method	Optimization Regio	n			
	Optimization Region Selection	Optimization Region				
-	Definition	^				
	Туре	Manufacturing Con	istraint			
	Subtype	Member Size				
	Suppressed	No				
-	Member Size					
	Minimum	Free				
	Maximum	Manual				
	Max Size	1.e-002 m				
	Gap Size	Manual				
	Value	1.e-002 m				
	1					




Local Strain Energy





- LSE is exposed as a new stress-norm alongside vonMises or Maximum Principal Stress.
- In some context LSE is equivalent to compliance.
- LSE permits to control the strain-energy of the body
 of interest

Details of "Response Constr	2"	▼ ‡ 🗆 ×
- Scope		
Scoping Method	Optimization Region	
Optimization Region Sele	Optimization Region 2	
- Definition		
Туре	Response Constraint	
Response	Global Stress	
Stress Type	Local Strain Energy	
Maximum	15 1	
Environment Selection	All Static Structural	
Suppressed	No	

 $\begin{cases} \min_{\{\Omega_1,\Omega_2,\Omega_3\}} complia \\ |vol < |vol$



For some context, the compliance is the strain energy of the whole model ...

 $\begin{cases} \min_{(\Omega_{1},\Omega_{2},\Omega_{3})} LSE \\ |vol < 50\% \end{cases}$



... so minimizing the total LSE (ie sum over all bodies) sometimes gives similar result to compliance.

 $\begin{cases} \min_{(\Omega_1, \Omega_2)} LSE \\ |vol < 50\%| \end{cases}$



By contrast, minimizing partial LSE (eg sum over 2 out of 3 bodies) can return non-intuitive design.









Improvement & Corrections

- Stress criterion
 - All elements lying in the Exclusion-Region are also excluded from the stress-control
 - A more accurate computation of the shape derivative has been implemented
 - Those changes may slightly affect the results with Level-set and Shape Optimization
- UDC modal: « Robust Frequency »
 - Consolidation of the feature in order to better manage the context of multiple modes
- Max member-size
 - A new numerical scheme has been devised permiting to better access the maximum thickness in the context of Shape optimization and in Discovery Live

Stress in prescribed-disp context











"Robust frequency" capability to manage multiple modes context



新科益工程仿真中心



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